

# Course 141: MECHANICS

## Problem Set 13: Solutions

### \* Problem 1

The radial energy equation of the particle is

$$E = T + V = \frac{m\dot{r}^2}{2} + \frac{J^2}{2mr^2} + \frac{k}{2r^2}$$

In the variable  $u = 1/r$ , we obtain

$$\frac{J^2}{2m} \left( \frac{du}{d\theta} \right)^2 + \frac{J^2}{2m} u^2 + \frac{k}{2} u^2 = E$$

and

$$J^2 \left( \frac{du}{d\theta} \right)^2 + (J^2 + mk) u^2 = 2mE$$

Solution of this equation is of the form  $u = \varepsilon \cos n(\theta - \theta_0)$ . Indeed, in this case  $\left( \frac{du}{d\theta} \right)^2 = \varepsilon^2 n^2 \sin^2 n(\theta - \theta_0)$  and identifying  $J^2 n^2 = J^2 + mk$  we can see that it is a solution of the radial energy equation if

$$\varepsilon^2 = \frac{2mE}{J^2 + mk}$$

### Orbits:

(a) Let  $J^2 + mk = 0$ . Then the radial equation reduces to

$$J^2 \left( \frac{du}{d\theta} \right)^2 = 2mE$$

The solution of this simple equation is  $u = \frac{\sqrt{2mE}}{J}(\theta - \theta_0) = 1/r$  and the orbit is given by  $r(\theta - \theta_0) = \frac{J}{\sqrt{2mE}}$ .

(b) Let  $J^2 + mk \equiv -C < 0$ . Then there are solutions with positive energy  $E > 0$  and the radial equation reads

$$J^2 \left( \frac{du}{d\theta} \right)^2 - Cu^2 = 2mE$$

The solutions are  $u = a \sinh n(\theta - \theta_0)$  where  $J^2 n^2 = J^2 + mk$  and  $a^2 = \frac{2mE}{J^2 + mk}$ , so the orbit is  $r \sinh n(\theta - \theta_0) = \sqrt{\frac{J^2 + mk}{2mE}}$

(c) Let  $J^2 + mk \equiv -C < 0$  and  $E = 0$ . The radial equation reads

$$J^2 \left( \frac{du}{d\theta} \right)^2 - Cu^2 = 0,$$

so, the solutions with zero total energy are  $u = ae^{\pm n\theta}$  where  $n^2 = C = J^2 + mk$  and the orbit is defined by the equation  $re^{\pm n\theta} = 1/a = \text{const.}$

(d) Let  $J^2 + mk \equiv -C < 0$  and  $E < 0$ . Then the radial equation is

$$J^2 \left( \frac{du}{d\theta} \right)^2 = -2mE$$

and the solution is  $u = a \cosh n(\theta - \theta_0)$ . To prove in one can apply the familiar relation  $\cosh^2 \theta - \sinh^2 \theta = 1$ .

\* **Problem 2**

(a) The potential energy of the central force  $\vec{F} = F(r)\hat{r}$  is

$$V(r) = - \int_{r_0}^r F(r') dr' = \int_{\infty}^r \frac{c}{r^{3/2}} = -\frac{2c}{\sqrt{r}}$$

If  $c > 0$ , the radial equation reads

$$E = \frac{m\dot{r}^2}{2} + \frac{J^2}{2mr^2} - \frac{2c}{r^{1/2}}$$

and the effective potential is  $U_{eff} = \frac{J^2}{2mr^2} - \frac{2c}{r^{1/2}}$ .

(b) The motion is unbounded for  $E \geq 0$ , it is bounded between  $r_1$  and  $r_2$  for  $E_0 < E < 0$  where  $E_0$  is the total energy on the circular orbit of radius  $r_0$ .

(c) Because

$$U'_{eff} = -\frac{J^2}{mr_0^3} + \frac{c}{r_0^{3/2}} = 0,$$

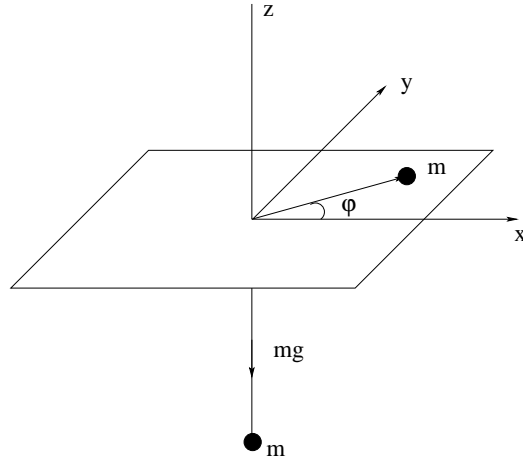
thus

$$r_0 = \left( \frac{J^2}{mc} \right)^{2/3}$$

For a circular orbit  $J = mvr_0$ , so the orbital period is

$$\tau = \frac{2\pi r_0}{v} = 2\pi \sqrt{\frac{m}{c}} r_0^{5/4}$$

\* **Problem 3**



(a) The potential energy of the system is

$$V = mgz = mg(r - l)$$

The kinetic energy of the mass on the table is

$$E_1 = \frac{m}{2}(\dot{r}^2 + r^2\dot{\phi}^2)$$

The kinetic energy of the suspended mass is

$$e_2 = \frac{m\dot{z}^2}{2} = \frac{m\dot{r}^2}{2}$$

Thus, the total energy of the system is

$$E = E_1 + E_2 + V = m\dot{r}^2 + \frac{mr^2\dot{\phi}^2}{2} + mg(r - l)$$

Conservation of the angular momentum  $\vec{J} = mr^2\dot{\phi}\hat{k}$  gives

$$E = m\dot{r}^2 + \frac{J^2}{2mr^2} + mg(r - l) = \text{const}$$

The effective radial equation then can be obtained by differentiation of this formula w.r.t. time:

$$\frac{dE}{dt} = 0 = \left( 2m\ddot{r} - \frac{J^2}{mr^3} + mg \right) \dot{r}$$

or, simple  $2m\ddot{r} - \frac{J^2}{mr^3} + mg = 0$ . This equation admits a solution for a circular orbit with  $\ddot{r} = 0, r = r_0 = \text{const}$ :

$$mg = \frac{J^2}{mr_0^3},$$

thus  $r_0 = \left( \frac{J^2}{m^2 g} \right)^{1/3}$ . Because of the conservation of the angular momentum  $\dot{\phi} = \frac{J}{mr_0^2} = \sqrt{\frac{g}{r_0}} = \text{const.}$

- (b) Because  $E = m\dot{r}^2 + U_{eff}$  where  $U_{eff} = \frac{J^2}{2mr^2} + mg(r - l)$  the physical motion is restricted by the condition  $U_{eff} \leq E$ . For  $E \geq \frac{J^2}{2mr^2}$  the motion is unbounded, i.e., both masses end up on the table. For  $U_{min} \leq E \leq \frac{J^2}{2mr^2}$  the motion is bounded. The minimum  $U_{min}$  corresponds to the stationary motion with  $r = r_0 = \text{const.}$
- (c) Expansion of the effective potential  $U_{eff}$  in Taylor series in the vicinity of the equilibrium distance  $r_0$  gives:

$$U_{eff}(r_0 + \delta) \approx U_{eff}(r_0) + \left. \frac{\partial U_{eff}}{\partial r} \right|_{r=r_0} \delta(t) + \frac{1}{2} \left. \frac{\partial^2 U_{eff}}{\partial r^2} \right|_{r=r_0} \delta^2(t)$$

The equation of motion yields  $\left. \frac{\partial U_{eff}}{\partial r} \right|_{r=r_0} = 0$  and the frequency of small oscillations is defined as  $\omega^2 = k/(2m)$  where

$$k = \left. \frac{\partial^2 U_{eff}}{\partial r^2} \right|_{r=r_0} = \frac{3J^2}{mr_0^4} = \frac{3mg}{r_0}$$

\* **Problem 4**

The energy of the planet before and after explosion is  $E_1 = T_1 + V_1$  and  $E_2 = T_2 + V_2$ , respectively. The kinetic energy conserves, i.e.,  $T_1 = T_2$  while the potential energy decreases as  $V_2 = V_1/2$  because  $V(r) = -\frac{GMm}{r}$  and  $M \rightarrow M/2$ .

On the other hand, for a circular orbit  $E_1 = \left| \frac{V_1}{2} \right|$ , thus  $T_1 = -\frac{V_1}{2} = T_2$  and after explosion

$$E_2 = -\frac{V_1}{2} + \frac{V_1}{2} = 0$$

That means the motion becomes parabolic.

\* **Problem 5**

The corresponding equation of motion is

$$m\ddot{\vec{r}} + k\vec{r} = 0$$

The solution of this isotropic oscillator problem is ( $\omega^2 = k/m$ )

$$\vec{r} = \vec{A} \cos \omega t + \vec{B} \sin \omega t$$

where  $\vec{A}, \vec{B}$  are arbitrary constant vectors which are determined by the initial conditions. Thus, the motion is periodic with the period  $\tau = 2\pi/\omega$  which is independent of the initial conditions.

For any fixed angle  $\theta$  we can rotate the vectors  $\vec{A}, \vec{B}$  as

$$\vec{A} = \vec{a} \cos \theta - \vec{b} \sin \theta; \quad \vec{B} = \vec{a} \sin \theta + \vec{b} \cos \theta$$

or, equivalently

$$\vec{a} = \vec{A} \cos \theta + \vec{B} \sin \theta; \quad \vec{b} = -\vec{A} \sin \theta + \vec{B} \cos \theta$$

The choice of the angle  $\theta$  is given by the condition that  $(\vec{a} \cdot \vec{b}) = 0$ , i.e.,  $\tan 2\theta = 2(\vec{A} \cdot \vec{B})/(A^2 - B^2)$ . Then the equation of motion becomes

$$x = a \cos(\omega t - \theta); \quad y = b \sin(\omega t - \theta); \quad z = 0$$

This gives the equation of the orbit

$$\frac{x^2}{a^2} + \frac{y^2}{b^2} = 1$$

which defines an ellipse with centre at the origin, and semi-axes  $a, b$ .